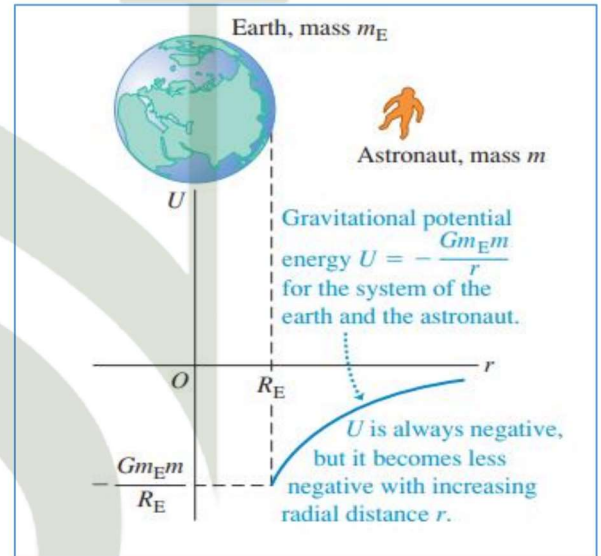
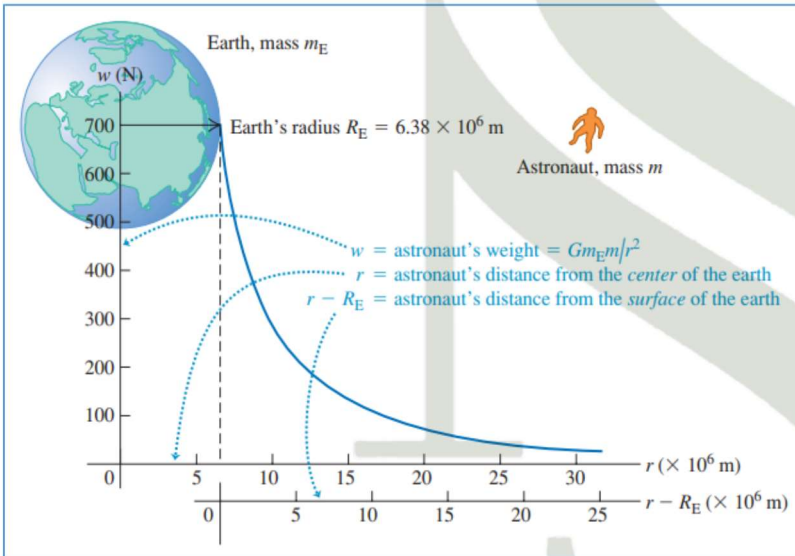
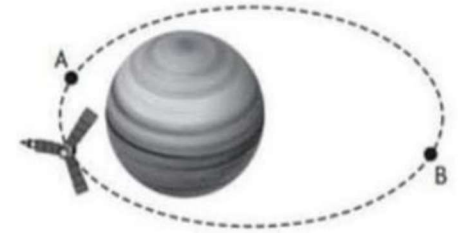


Exam Model Questions:

1. You wish to put a 1000 kg satellite into a circular orbit 300 km above the earth's surface.
 - a. What speed, period and radial acceleration will it have?
 - b. How much work must be done to the satellite to put it in orbit?
 - c. How much additional work would have to be done to make the satellite escape the earth.

The earth radius and mass are $R = 6.37 \times 10^6\text{ m}$ and $M = 5.97 \times 10^{24}\text{ kg}$.

2. a. Juno is a NASA orbiter with a mission to survey Jupiter. It is in an elliptical orbit around Jupiter as shown in the figure below. The gravitational potential at point A in the orbit of Juno is $-1.70 \times 10^9\text{ J kg}^{-1}$.
 - i. State what is meant by a gravitational potential at point A is $-1.70 \times 10^9\text{ J kg}^{-1}$.
 - ii. At point B, Juno is $1.69 \times 10^8\text{ m}$ from the centre of Jupiter. If the mass of Jupiter is $1.90 \times 10^{27}\text{ kg}$, calculate the gravitational potential at point B.
- b. Draw graphs showing the variation of acceleration due to gravity with (i) height (ii) depth from the surface of the earth.



Orbits: When a satellite moves in a circular orbit, the centripetal acceleration is provided by the gravitational attraction of the earth. Kepler's three laws describe the more general case: an elliptical orbit of a planet around the sun or a satellite around a planet. (See Examples 13.6–13.9.)

$$v = \sqrt{\frac{Gm_E}{r}} \quad \text{(speed in circular orbit)} \quad (13.10)$$

$$T = \frac{2\pi r}{v} = 2\pi r \sqrt{\frac{r}{Gm_E}} = \frac{2\pi r^{3/2}}{\sqrt{Gm_E}} \quad \text{(period in circular orbit)} \quad (13.12)$$

